

# FY23 Strategic Initiatives Research and Technology Development (SRTD)

# Additively Manufactured Rover Chassis with Integrated Thermal Control for **Extreme Cold Environments**

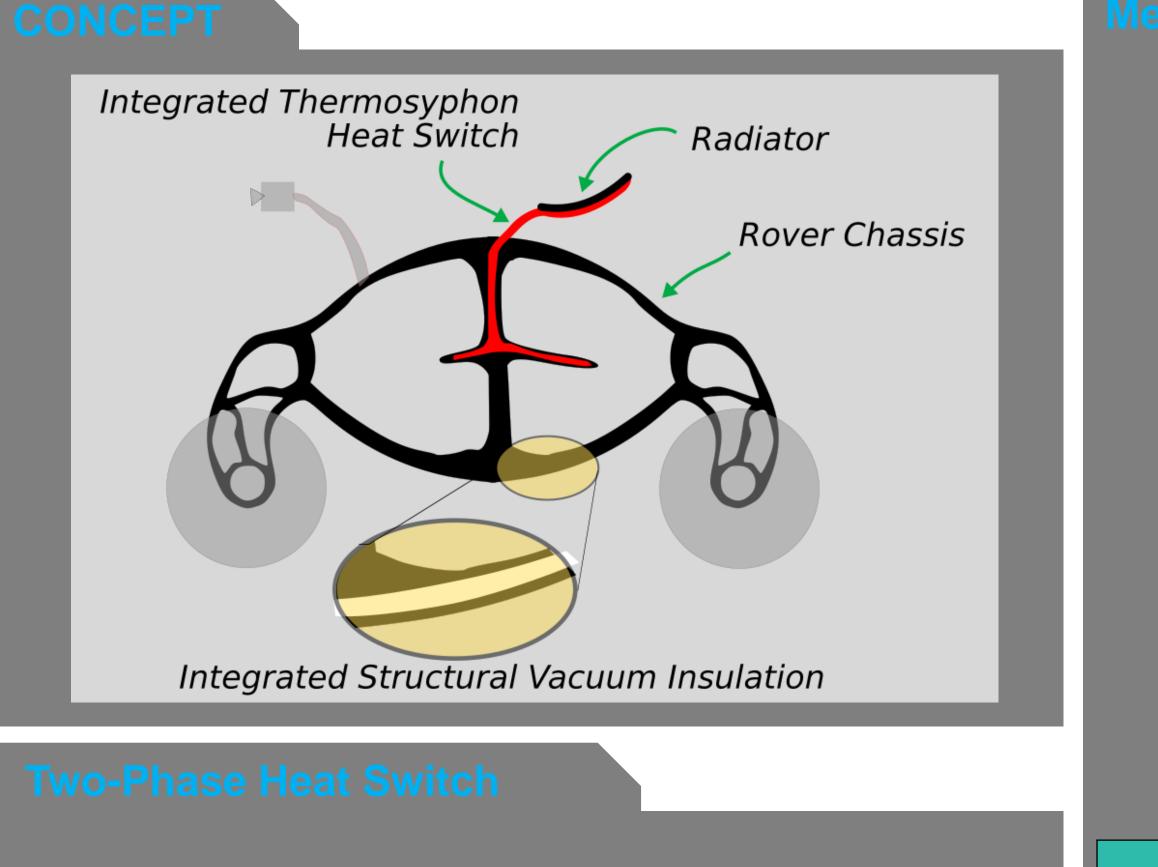
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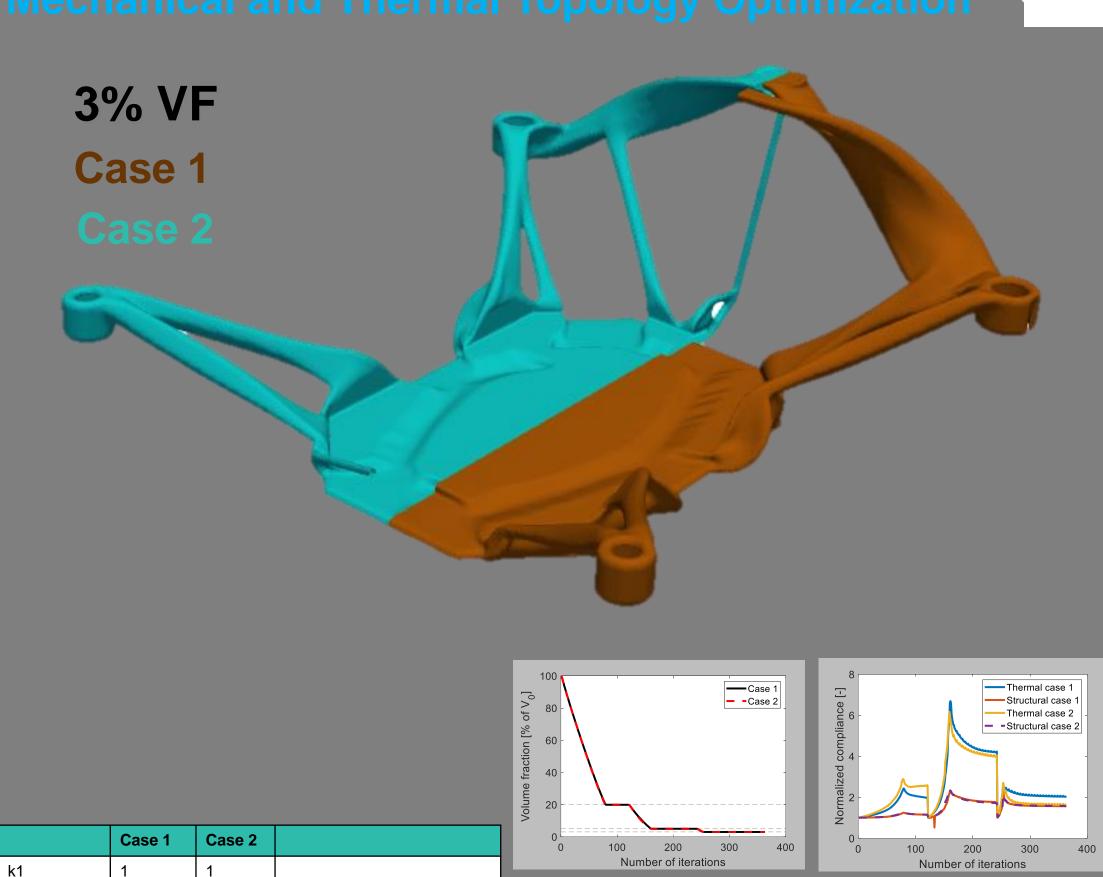
Strategic Focus Area: Lunar Science/ Moon and Mars Extreme Cold, Steep Terrain Rover | Strategic Initiative Leader: John D Baker

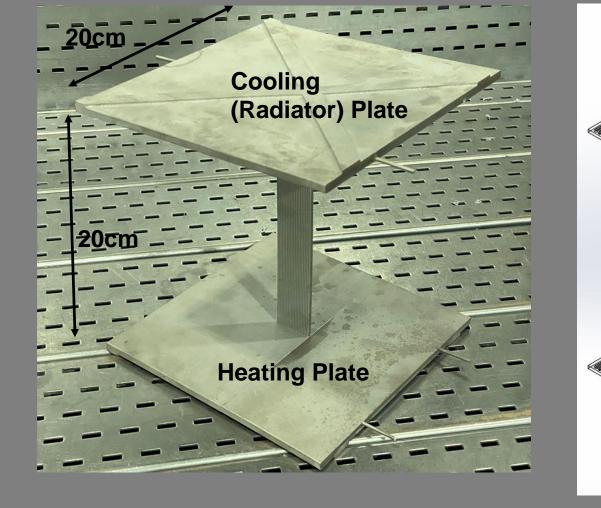
## **Objectives:**

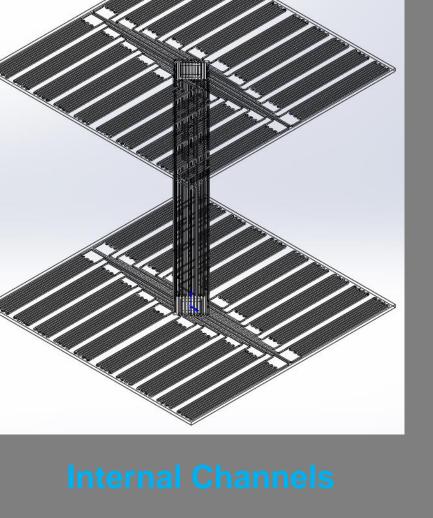
The goal of this three-year strategic RTD task is to develop an topologically optimized additively manufactured (AM) rover chassis that incorporates a heat switch and insulation elements to enable new mission concepts to the Moon and Mars. The rover chassis is a mutlifunctional thermal-structural component. This year was the second year of the task

and the high-level goals were to mature the basic thermal technology elements (AM heat switch and AM thermal insulation), as well as develop a notional rover chassis design. The final goal will be to qualify at TRL 5 an AM rover chassis that has been thermally/structurally optimized and incorporates a heat switch and insulating elements.

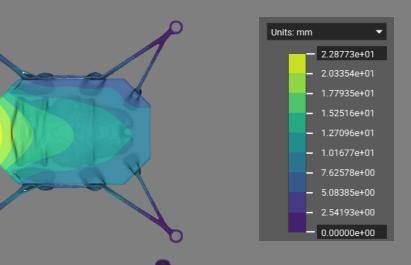




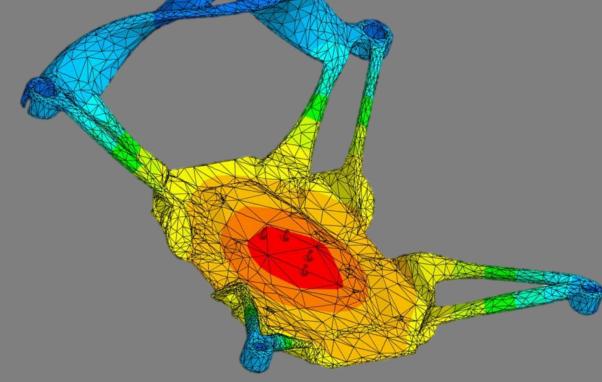




l	k2	-0.01	-0.03		$ \begin{array}{ll} \min & k_1 \frac{C_S}{C_{S0}} + k_2 \frac{C_T}{C_{T0}} \\ s.t. & V \leq VF \end{array} \end{array} $	$C_S$ is the structural compliance $\equiv$ strain energy	
	Level 0 - VF	20 %	20%	FEM element size = 16 mm		$\Gamma \Gamma $	$C_T$ is the thermal compliance $\equiv$ thermal energy
	Level 1 - VF	5 %	5%	FEM element size = 10 mm			
	Level 2 - VF	3 %	3%	FEM element size = 8 mm		$V \leq VF$	







## **National Aeronautics and Space Administration**

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